62790/94 MBER PCT/AU94/00124



		,ACI)
(51) International Patent Classification 5:		(11) International Publication Number: WO 94/20809
F41A 21/00, 21/06	A1	(43) International Publication Date: 15 September 1994 (15.09.94)
(21) International Application Number: PC	CT/AU94/001	(81) Designated States: AT, AU, BB, BG, BR, BY, CA, CH, CN, CZ, DE, DK, ES, FI, GB, GE, HU, JP, KG, KP, KR,
(22) International Filing Date: 14 March	1994 (14.03.9	
		TE TE LIL ACC ME BT SEL CARE STORE (BE BI CE

(30) Priority Data:

PL 7773	12 March 1993 (12.03.93)	ΑU
PL 8876	19 May 1993 (19.05.93)	AU
PM 1201	15 September 1993 (15.09.93)	AU
PM 2868	9 December 1993 (09.12.93)	ΑÜ
PM 3314	12 January 1994 (12.01.94)	AU
	•	

(71)(72) Applicant and Inventor: O'DWYER, James, Michael [AU/AU]; 1256 Ross River Road, Kelso, Townsville, QLD 4814 (AU).

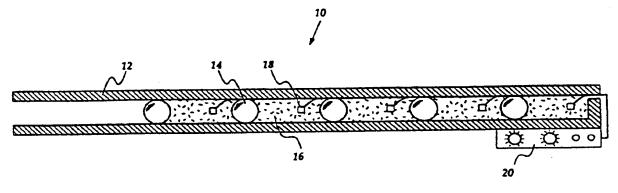
(74) Agent: PIZZEY, John, Kingston; Pizzey & Company, 6/444 Queen Street, Brisbane, QLD 4000 (AU).

CG, CI, CM, GA, GN, ML, MR, NE, SN, TD, TG).

#### **Published**

With international search report. Before the expiration of the time limit for amending the claims and to be republished in the event of the receipt of amendments.

(54) Title: A BARREL ASSEMBLY



#### (57) Abstract

A barrel assembly (10) includes barrel (12), projectile assemblies (14) axially disposed within the barrel, propellant charges (16), ignition means (18), and control means (20). Projectile assemblies (14) are sequentially fired from the barrel. Also disclosed is an arming system, a barrel construction, means for preventing compression of the propellant charges and means for ensuring sealing between the projectile assembly and the barrel.

#### "A BARREL ASSEMBLY"

#### -- BACKGROUND OF THE INVENTION --

The invention relates to firearms.

The invention has utility as an automatic, high rate of fire, firearm whereby it may be used for example, as a close-in ship-board defense against bombs, missiles or attack aircraft for launching large numbers of projectiles within a short period of time. The invention also has utility in hand guns such as a rapid fire pistol or rifle which may be 10 disposable.

Currently, most firearms use cartridge ammunition which is mechanically fed to a barrel. Such firearms have numerous moving parts, tend to be heavy and complex, may jamb or be unreliable, and require elaborate delivery and loading

15 systems to support the rate of fire. The rate of fire of automatic firearms of this type is limited by the time required to load the cartridge, seal the barrel, unseal the barrel and eject the empty case.

More recently, firearms have begun to utilise caseless
20 ammunition which obviates the need to eject an empty case
subsequent to firing. However, these firearms retain many of
the problems of conventional firearms.

#### -- SUMMARY OF THE INVENTION --

The present invention aims to provide an alternative 25 system which will alleviate at least one of the disadvantages of the prior art.

According to one aspect this invention provides a barrel assembly including:-

a barrel;

a plurality of projectile assemblies axially disposed within the barrel for operative sealing engagement with the bore of the barrel;

discrete propellant charges for propelling respective projectile assemblies sequentially through the muzzle of the 35 barrel;

ignition means for igniting the discrete propellant charges; and

control means for selectively and sequentially actuating the ignition means.

The ignition means may be electrical, chemical, mechanical or any other conventional primer. Conveniently, the ignition means is electrical and the control means is an electrical control adapted to provide electrical ignition pulse to the respective ignition means. Suitably the control means is configured to enable a user to selectively control the rate, number, and frequency of the pulses to provide a desired firing pattern. The control means may fire the projectile assemblies singly, in pairs, or in any other combinations.

The projectile assembly may be round, conventionally shaped or dart-like and the fins thereof may be off-set to generate a stabilising spin as the dart is propelled from a barrel which may be a smooth-bored barrel. In addition the barrel assembly may find utility as a removable/replaceable 20 barrel of a rifle or pistol.

Alternatively the barrel assembly constitutes one of a plurality of barrel assemblies and the control means may actuate the ignition means of each of the barrel assemblies in such manner that a sequential plurality of arrays of 25 projectile assemblies are propelled in following relationship. Aiming and firing of the arrays of projectile assemblies may be controlled by a conventional radar fire control system or other known fire control systems. The individual barrel assemblies may be aimed such that the array of projectile assemblies converges at a particular range to give a maximum density of projectile assemblies at that range.

Alternatively, the array of projectile assemblies may diverge to maximise coverage of an area. Thus, the average 35 separation distance at the target between the projectile

3

assemblies in an array can be predetermined and adjusted to suit the nature and range of the target. Of course, the individual barrel assemblies may be fired randomly or independently of the other barrel assemblies.

The plurality of projectile assemblies may be disposed in a continuous abutting relationship throughout the barrel either by the projectile assemblies abutting one another or abutting column means intermediate the projectile assemblies to form a compression resistant column able to resist compression of the projectile assemblies or propelling charges associated therewith due to pressure generated by the firing of the leading projectile assemblies.

The propelling charges may be either solid or granular and compression of either may be an undesirable, moreover, 15 movement of the projectile assemblies relative to the barrel may cause misalignment of the ignition means with their respective propellant charges.

It is preferred that the ignition means be disposed at the leading end of the propellant charge so as to minimise 20 possible energy loss in accelerating the front portion of the propellant charge.

It is preferred that each projectile assembly includes a projectile head and extension means for at least partly defining a propellant space. Preferably, the extension means 25 includes a spacer assembly which extends rearwardly from the projectile head and abuts an adjacent projectile assembly.

In one embodiment, the spacer assembly extends through the propellant space and the projectile head whereby compressive loads are transmitted directly through abutting 30 adjacent spacer assemblies. In such embodiment the spacer assembly may add support to the extension means which may be a thin cylindrical rear portion of the projectile head. Furthermore the extension means may form an operative sealing contact with the bore of the barrel to prevent burn leakage 35 past the projectile head.

It is preferred that the spacer assembly includes a rigid collar which extends outwardly to engage a thin cylindrical rear portion of the malleable projectile head in operative sealing contact with the bore of the barrel such 5 that axially compressive loads are transmitted directly between spacer assemblies thereby avoiding deformation of the malleable projectile head.

In another embodiment, complementary wedging surfaces are disposed on the spacer assembly and projectile head 10 respectively whereby the projectile head is urged into engagement with the bore of the barrel in response to relative axial compression between the spacer means and the projectile head. In such arrangement the projectile head and spacer assembly may be loaded into the barrel and thereafter 15 an axial displacement is caused to ensure good sealing between the projectile head and barrel. Suitably the extension means is urged into engagement with the bore of the barrel.

Preferably, the projectile head defines a tapered
20 aperture at its rearward end into which is received a
complementary tapered spigot disposed on the leading end of
the spacer assembly, wherein relative axial movement between
the projectile head and the complementary tapered spigot
causes a radially expanding force to be applied to the
25 projectile head.

The barrel may be non-metallic and the bore of the barrel may include recesses which may fully or partly accommodate the ignition means. In this situation the barrel houses electrical conductors which facilitate electrical 30 communication between the control means and ignition means. This arrangement may be utilised for disposable barrel assemblies which have a limited firing life and the ignition means and control wire or wires therefor can be integrally manufactured with the barrel.

35 In an alternative arrangement, a barrel assembly

electrodes.

5

includes ignition apertures in the barrel and the ignition means are disposed outside the barrel and adjacent the apertures. The barrel may be surrounded by a non-metallic outer barrel which may include recesses adapted to

5 accommodate the ignition means. The outer barrel may also house electrical conductors which facilitate electrical communication between the control means and ignition means. The outer barrel may be formed as a laminated plastics barrel which may include a printed circuit laminate for the ignition means.

Both of the above arrangements lend themselves to a modular or disposable construction. The barrel assemblies may be adapted for firing as is, or may be adapted for mounting within a housing.

For safety, the barrel assembly may include an arming switch associated with each ignition means which is closed in response to the preceding projectile assembly being discharged. Preferably, the arming switch is closed by biasing means which are normally resisted by the preceding 20 projectile assembly. In a preferred embodiment, the projectile head and spacer assembly each constitute switch contacts which are normally electrically isolated from each other and wherein an electrical circuit between the barrel and spacer body is completed in response to the preceding 25 projectile assembly being discharged. In this arrangement, the barrel, which is in electrical contact with the projectile head, is also in contact with one of the

In a further aspect this invention resides broadly in a 30 method of defending an airspace, including:-

providing a plurality of barrel assemblies substantially as defined above, and

sequentially igniting propellant charges in the barrel assemblies in rapid succession to propel sequential arrays of 35 projectile assemblies into the airspace.

#### -- BRIEF DESCRIPTION OF THE DRAWINGS --

In order that this invention may be more readily understood and put into practical effect, reference will now be made to the accompanying drawings which illustrate 5 typical embodiments of the invention and wherein:-

- FIG. 1 is a sectional and schematic view of an embodiment of a barrel assembly according to the invention:
- FIG. 2 schematically illustrates the concept of a plurality of barrel assemblies according to the invention being massed in pods;
  - FIG. 3 is a schematic view of arrays of projectile assemblies being fired from the pods of FIG. 2;
  - FIG. 4 is a sectional and schematic view of an embodiment of a barrel assembly according to the
- embodiment of a barrel assembly according to the invention wherein the projectile assembly is in the form of a dart;
  - FIG. 5 is a sectional and schematic view of another embodiment of a barrel assembly according to the
- 20 invention;
  FIG. 6 is a sectional and schematic view of another embodiment of a barrel assembly according to the
  - invention;
    FIG. 7 is a sectional and schematic view of another
- embodiment of a barrel assembly according to the invention;
  - FIG. 8 is a sectional and schematic view of another embodiment of a barrel assembly according to the invention;
- 30 FIG. 9 is a sectional and schematic view of another embodiment of a barrel assembly according to the invention;
  - FIG. 10 is a sectional and schematic view of another embodiment of a barrel assembly according to the
- 35 invention;

7

FIG. 11 is a diagrammatic representation of a pistol made in accordance with the present invention, and FIGS. 12 and 13 illustrate an alternate form of projectile.

-- DESCRIPTION OF THE PREFERRED EMBODIMENT --Referring to FIG. 1, there is illustrated a barrel assembly 10 including a barrel 12, a plurality of spherical projectiles 14 axially disposed within barrel 12 for operative sealing engagement with the bore of barrel 12, 10 discrete propellant charges 16 disposed between adjacent projectile assemblies 14 for propelling the respective projectile assemblies 14 individually and sequentially through the muzzle of barrel 12, ignition means 18 for igniting discrete propellant charges 16, and control means 20 15 for selectively and sequentially actuating ignition means 18.

In use, the leading projectile assembly 14 is propelled in response to ignition of the leading propellent charge 16 by the leading ignition means 18. Thereafter the following projectile assemblies are sequentially propelled in like There is no ammunition delivery system or moving parts, and the firing rate is practically limited only by the time taken for each projectile assembly to exit the barrel.

The control means may have time delay means to control the rapidity of fire and or timing means permitting a 25 selected number of sequential ignitions in response to each manual actuation of the ignition means, such as by squeezing a trigger. A mode switch may be associated with the control means to enable a user to select the form of firing, ie full barrel discharge, short bursts of rapid fire, sequential fire 30 of a selected number of projectiles, single shot firing per Integrated circuit electronic control means actuation etc. are preferably utilised as the control means and may be manufactured as part of the barrel assembly.

Referring to FIG. 2, the barrel assembly constitutes one 35 of a plurality of barrel assemblies and the control means

actuates the ignition means of each of the barrel assemblies in such manner that a sequential plurality of arrays of projectile assemblies are propelled in following relationship as shown in FIG. 3. The plurality of barrel assemblies forms 5 a pod 22 and a plurality of pods are mounted on a trainable mount 24. The aiming and firing of the barrel assemblies is controlled by a radar fire control system 25 or other conventional system.

In one form, each barrel is 2.25 metres long and has an 10 outside diameter of 20 mm. The combined propelling charge/projectile assembly length is 50 mm. Leaving 0.25 metres of the barrel free, 40 projectile assemblies together with their associated propellent charges can be pre-loaded into the barrel. The pod has a cross-sectional dimension of 15 0.75 metres by 0.75 metres for example and therefore accommodates approximately 1200 barrel assemblies. Thus, a pod can be pre-loaded with 48000 projectile assemblies.

This enables significant fire-power to be associated with a relatively small weapon and a very high discharge rate 20 to be achieved, bearing in mind the firing rate of each individual barrel assembly may be significantly in excess of the rate achievable by conventional automatic firearms. barrel assemblies may be formed as a relatively lightweight honeycomb structure which will be very stiff and if desired 25 the barrels may be arranged to focus at a point relatively close to the weapon with a view to counteracting the spreading tendencies produced by the expansion of the hot explosion gases radiating in an outwards direction. Alternatively a box-like baffle could be used to prevent the 30 immediate outward spread of the gases. This baffle may be slidably supported about the outer barrel section for extension past the end of the barrels during firing. A further manner of alleviating this perceived effect would be to slightly stagger the firing of the projectiles.

Referring to the embodiments of FIGS. 4 to 10,

projectile assemblies 14 are disposed in axial abutting relationship to form a compression resistant column. Axially compressive loads are created by the pressures generated in the barrel by the propulsion of preceding projectile 5 assemblies. Compression can result in an alteration of the burn rate of a propelling charge, misalignment of ignition means with respective propelling charges or even premature ignition of propelling charge.

Each projectile assembly 14 includes a projectile head 10 26 and means for defining a propellant space in the form of spacer assembly 28 which extends axially and rearwardly from projectile head 26 and abuts an adjacent projectile assembly 14.

Projectile head 26 is formed from a heavy malleable 15 material such as lead to facilitate operative sealing with barrel 12, and spacer assembly 28 is formed of a rigid material such as steel.

In the embodiment of FIG. 5, the spacer assembly 28 takes the form of a cylinder axially extending from 20 projectile head 26. The interior of the cylinder accommodates propellent charge 16 and is structurally reinforced to prevent excessive radial expansion. The end of the cylinder is adapted to abut the leading end of the subsequent projectile assembly 14.

25 Referring to the embodiments of FIGS. 6 and 7, spacer assembly 28 extends through projectile head 26 to the leading end of projectile head 26 whereby compressive loads are transmitted directly between adjacent spacer assemblies 28. Spacer assembly 28 supports a thin cylindrical rear portion 30 30 of projectile head 26 in operative sealing contact with the bore of barrel 12. Specifically, spacer assembly 28 includes a radially outwardly extending collar flange 32 which supports thin cylindrical rear portion 30 of projectile head 26 in operative sealing contact with the bore 35 of barrel 12.

Referring to the embodiments of FIGS. 9 and 10, complementary wedging surfaces 34, 36 are disposed on spacer assembly 28 and projectile head 26 respectively whereby thin cylindrical rear portion 30 of projectile head 26 is urged 5 into engagement with the bore of barrel 12 in response to an axially compressive load being applied to projectile assembly 14. Projectile head 26 defines a tapered aperture 38 at its rearward end into which is received a complementary tapered spigot 40 disposed on the leading end of spacer assembly 28. 10 Relative axial movement between tapered aperture 38 and complementary tapered spigot 40 causes a radially expanding force to be applied to thin cylindrical rear portion 30 of projectile head 26.

In the embodiment of FIG. 7, barrel 12 is non-metallic 15 and the bore of the barrel includes recesses 42 which at least partly accommodate ignition means 18. Barrel 12 may be formed of kevlar, carbon fibre, glass reinforced polymer or the like. Thus, the barrel assembly may be lightweight and disposable. Barrel 12 houses electrical conductors 44 which 20 facilitate electrical communication between the control means and ignition means.

In the embodiments of FIGS. 8 and 9, barrel 12 includes ignition apertures 46 and ignition means 18 are disposed outside the barrel and adjacent the apertures. Barrel 12 is 25 surrounded by a non-metallic outer barrel 48, the bore of the outer barrel including recesses adapted to at least partly accommodate the ignition means. The barrel assembly may be slidably received in sheath 50. Outer barrel 48 houses electrical conductors 44 which facilitate electrical 30 communication between the control means and ignition means 18.

Referring to FIG. 10, arming switch 52 associated with ignition means 18 is closed in response to the preceding projectile assembly being discharged. Specifically, arming 35 switch is closed by biasing means 54 once the preceding

11

projectile assembly has been propelled. Projectile head 26 and spacer assembly 28 each constitute switch contacts which are normally electrically isolated from each other by insulating layer 56. An electrical circuit between barrel 12 and spacer assembly 28 is completed when arming switch 52 closes in response to the preceding projectile assembly being discharged. The ignition means 18 is thus armed only when the preceding projectile assembly has been discharged.

A four barrel hand gun 60 is illustrated in FIG. 11.

10 The barrels of the four barrel set 61, are arranged in a square formation, and are fed by a matching replaceable four barrel magazine block 62 which slots into a cutout 63 at the base of the barrel set 61. The barrel set 61 is formed integrally with the handgrip 64 which contains the electronic 15 controls for the ignition means.

The four barrel magazine block 62 is loaded with 5 rounds per barrel, which number may of course be varied depending on the size of the block and the size of the round. In this embodiment the magazine block 62 contains twenty 20 rounds.

A variable fire rate and pattern switch 66, is provided for selectively controlling the electronic ignition circuits within the magazine block 62 which connect electrically with the circuits in the hand gun via contacts which meet when the 25 magazine block 62 is slid into position. The switch 66 may be adjusted for electronic control to enable a user to fire individual rounds with each action of the trigger 65, up to four rounds simultaneously, or all rounds automatically on all barrels. A safety catch 68 may also be provided for 30 electrically disabling the weapon. Preferably the cartridges are disposable and may be provided in different formats so that a user may select and/or quickly change the type of rounds to be fired.

The projectiles for use with the above described 35 embodiments may be provided with external flights or spiral

ridges as illustrated in FIGS. 12 and 13. The ridges 70 are provided on the nose of the projectile to impart spin during flight. In the form illustrated a 7.62mm bullet 71 has four spiraling ridges 70 radiating from the nose of the bullet.

- 5 The ridges are of an average height of 1.5mm and extend the length of the nose of the bullet, but not along the side of the bullet. The pitch is suitably formed as to provide a single revolution of the bullet about its longitudinal axis for every meter travelled.
- Of course two or more spiraling ridges, spaced evenly around the bullet nose may be utilised if desired. Furthermore the height of the ridges, the length of the ridges, the pitch or degree of spiraling, the geometric curve form of the spiral, may be varied to suit the desired flight characteristics. The ridges may also extend along the side of the bullet. The cross section profile of the spiral ridges may be relatively flat, or steep according to the intended use of the ammunition, and the desired degree of reaction to the airflow.
- As illustrated in FIG. 13, the ridges 70 may have a steep leading face 72, which offers resistance to the airflow over the bullet, and causes the bullet to rotate, a flat top portion 73 and trailing faces 74 which slope gently to the surface of the bullet.
- Such ammunition may also be used in rifled barrel weapons to advantage. Also as the spirals on the bullet would assist in producing the spin during firing, the normal pressure applied by the edge of the rifling lands against the soft metal of the bullet would be reduced. Therefore the
- 30 bullet would not require the rifling to cut as long a track along the side of the bullet. Rather, the small expanding band of the Minie' gas sealing system would then be adequate to assist with spin acceleration. On impact with soft targets, the spiral bullet of the present invention would
- 35 tend to react to the increased pressure on the ridges by

13

maintaining a high rate of twist, as it progresses through the target material.

It will of course be realised that the above has been given only by way of illustrative example of the invention, 5 and that all such modifications and variations thereto as would be apparent to persons skilled in the art are deemed to fall within the broad ambit and scope of the invention as is herein set forth.

#### -- CLAIMS --

- 1. A barrel assembly including:
  - a barrel;
- a plurality of projectile assemblies axially disposed 5 within the barrel for operative sealing engagement with the bore of the barrel;

discrete propellant charges for propelling respective projectile assemblies sequentially through the muzzle of the barrel;

ignition means for igniting the discrete propellant charges; and

control means for selectively and sequentially actuating the ignition means.

- 2. A barrel assembly as defined in claim 1, wherein the 15 barrel assembly constitutes one of a plurality of barrel assemblies and the control means actuates the ignition means of each of the barrel assemblies in such manner that a sequential plurality of arrays of projectile assemblies are propelled in following relationship.
- 20 3. A barrel assembly as defined in claim 1 or 2, wherein the plurality of projectile assemblies are disposed in axial abutting relationship to form a compression resistant column.
- 4. A barrel assembly as defined in claim 3, wherein each projectile assembly includes a projectile head and extension 25 means for at least partly defining a propellant space.
  - 5. A barrel assembly as defined in claim 4, wherein the extension means includes a spacer assembly which extends axially from the projectile head and abuts an adjacent projectile assembly.

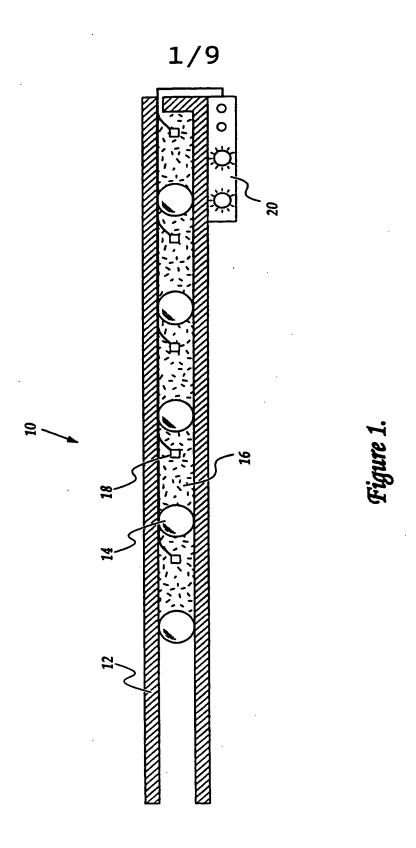
- 6. A barrel assembly as defined in claim 5, wherein the spacer assembly extends through the projectile head.
- 7. A barrel assembly as defined in claim 5 or 6, wherein the spacer assembly supports a thin cylindrical rear portion of 5 the projectile head in operative sealing contact with the bore of the barrel.
- 8. A barrel assembly as defined in claim 7, wherein the spacer assembly includes a radially outwardly extending collar which maintains the thin cylindrical rear portion of 10 the projectile head in operative sealing contact with the bore of the barrel.
  - 9. A barrel assembly as defined in claim 8, wherein the projectile head is formed from a heavy malleable material and the spacer assembly is formed of a rigid material.
- 15 10. A barrel assembly as defined in claim 5, wherein complementary wedging surfaces are disposed on the spacer assembly and projectile head respectively.
- 11. A barrel assembly as defined in claim 10, wherein the projectile head and spacer assembly are loaded into the 20 barrel and thereafter an axially compressive load is applied to ensure good sealing between the projectile head and barrel.
- 12. A barrel assembly as defined in claim 11, wherein the projectile head defines a tapered aperture at its rearward 25 end into which is received a complementary tapered spigot disposed on the leading end of the spacer assembly, wherein relative axial movement between the tapered aperture and the complementary tapered spigot causes a radially expanding force to be applied to the projectile head.

- 13. A barrel assembly as defined in any preceding claim, wherein the barrel is non-metallic and the bore of the barrel includes recesses which at least partly accommodate the ignition means, and wherein the barrel houses electrical conductors which facilitate electrical communication between the control means and ignition means.
- 14. A barrel assembly as defined in any one of claims 1 to 12, wherein the barrel includes ignition apertures and wherein the ignition means are disposed outside the barrel 10 and adjacent the apertures.
- 15. A barrel assembly as defined in claim 14, wherein the barrel is surrounded by a non-metallic outer barrel, the bore of the outer barrel including recesses adapted to at least partly accommodate the ignition means, and wherein the outer barrel houses electrical conductors which facilitate electrical communication between the control means and ignition means.
- 16. A barrel assembly as defined in any one of claims 5 to 15, wherein the barrel assembly includes an arming switch 20 associated with each ignition means which is closed in response to the preceding projectile assembly being discharged.
- 17. A barrel assembly as defined in claim 16, wherein the arming switch is closed by biasing means which are normally 25 resisted by the preceding projectile assembly.
  - 18. A barrel assembly as defined in claim 16 or 17, wherein the projectile head and spacer assembly each constitute switch contacts which are normally electrically isolated from each other and wherein an electrical circuit between the

barrel and spacer body is completed in response to the preceding projectile assembly being discharged.

- 19. A barrel assembly substantially as herein described with reference to any one of the accompanying Figures.
- 5 20. A method of defending an airspace, including:providing a plurality of barrel assemblies as defined in
  any one of the preceding claims, and

sequentially igniting propellant charges in the barrel assemblies in rapid succession to propel sequential arrays of 10 projectile assemblies into the airspace.



2/9

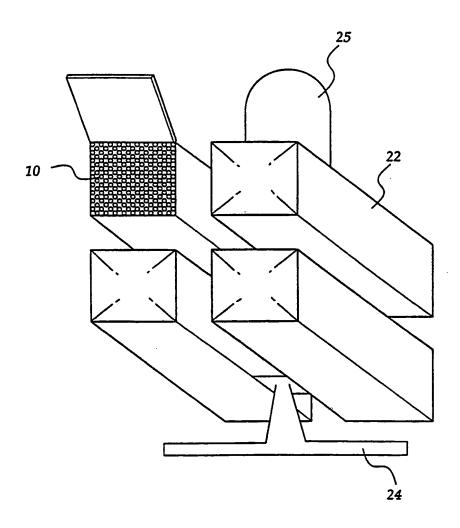
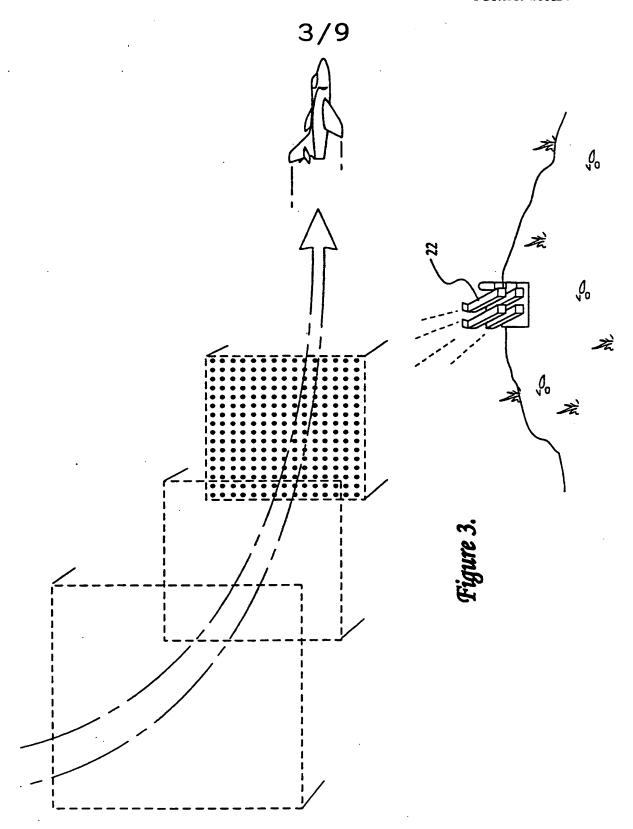


Figure 2.



SUBSTITUTE SHEET (Rule 26)

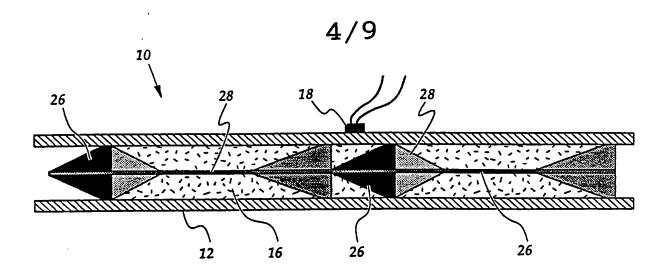


Figure 4.

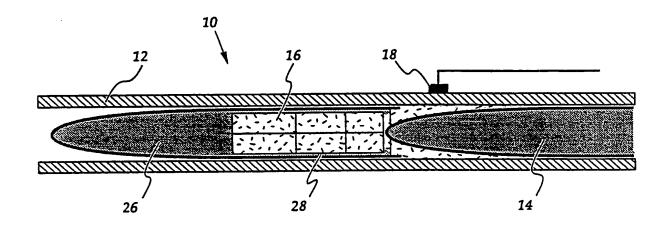


Figure 5.

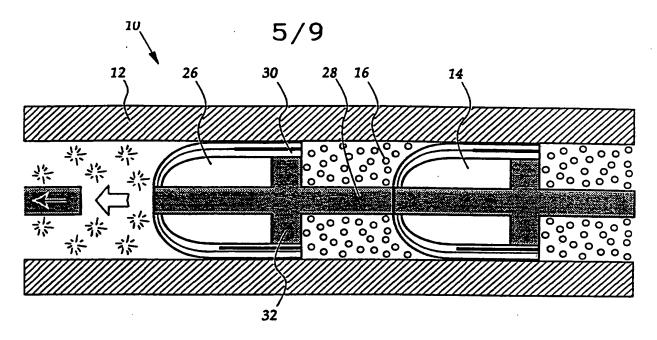


Figure 6.

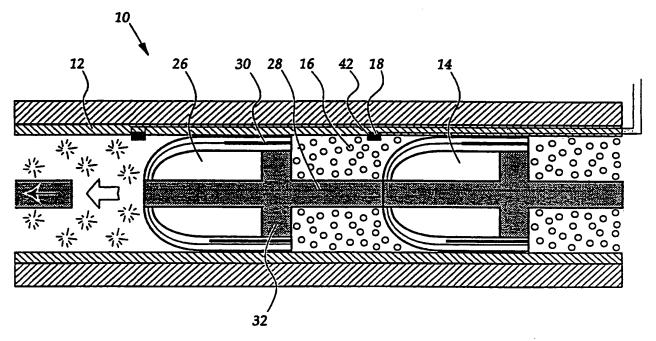


Figure 7.

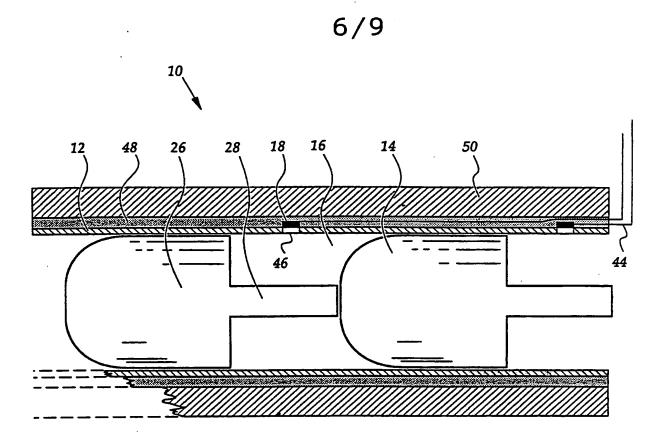


Figure 8.

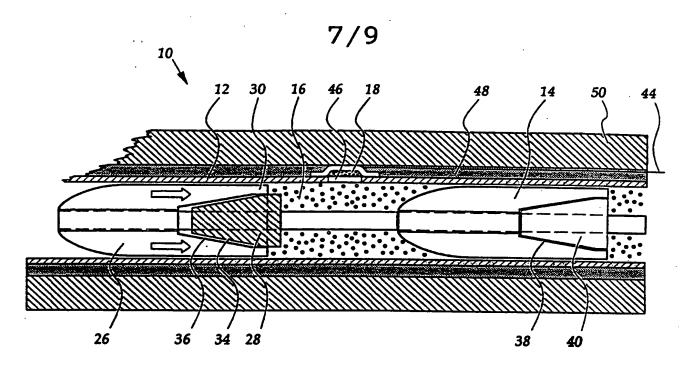


Figure 9.

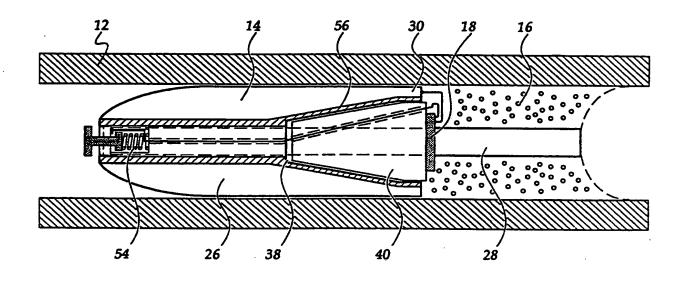


Figure 10.

8/9

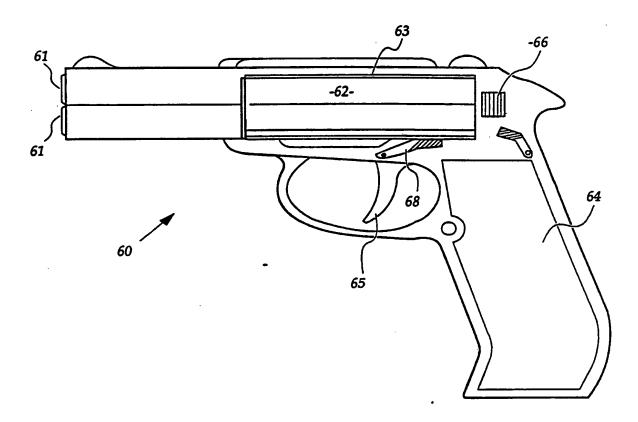
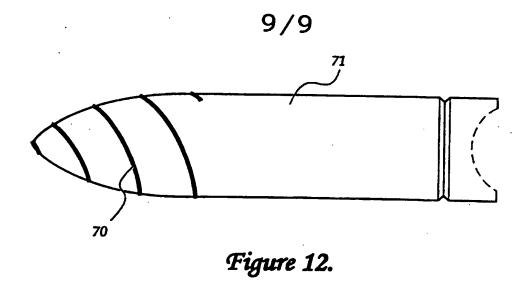
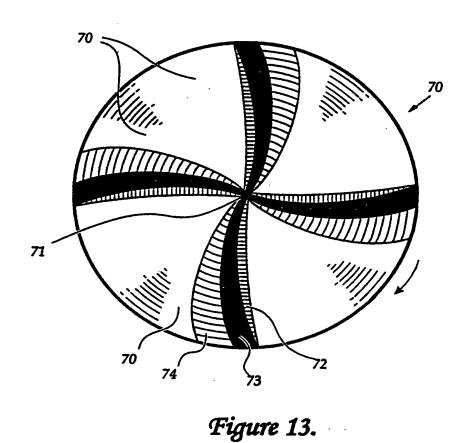


Figure 11.





SUBSTITUTE SHEET (Rule 26)





International application No. PCT/AU 94/00124

A. CLASSIFICATION OF SUBJECT MATTER Int. Cl. <sup>5</sup> F41A 21/00, 21/06				
According to International Patent Classification (IPC) or to both national classification and IPC				
В. 1	FIELDS SEARCHED			
Minimum documentation searched (classification system followed by classification symbols) IPC F41A 21/00, 21/06, 9/72, F41C 3/00, 7/00, 21/00, 21/06, F41D 7/00, F41F 1/00, 1/08, 17/00				
Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched AU: IPC as above Aust Classification 89.3				
Electronic data base consulted during the international search (name of data base, and where practicable, search terms used) WPAT JAPIO				
C.	DOCUMENTS CONSIDERED TO BE RELEVA	NT		
Category*	Citation of document, with indication, where a	ppropriate, of the relevant passages	Relevant to Claim No.	
A	DE, A, 314073 (DELLING) 9 August 1919 (09.08.19) page 1 lines 16-60 1-20			
A	FR.A. 1537857 (SOCIETE D'ETUDE DE LA PROPULSION PAR REACTION) 22 July 1968 (22.07.68) page 2 lines 19-32 1-20			
A	GB,A, 2161908 (GUTHRIE) 22 January 1986 (22.01.86) see abstract 1-20			
X Further in the	er documents are listed continuation of Box C.	X See patent family annex		
* Special categories of cited documents:  "A" document defining the general state of the art which is not considered to be of particular relevance earlier document but published on or after the international filing date document which may throw doubts on priority claim(s) or which is cited to establish the publication date of another citation or other special reason (as specified) document referring to an oral disclosure, use, exhibition or other means document published prior to the international filing date but later than the priority date claimed  "O" document published prior to the international filing date but later than the priority date claimed  "O" document published prior to the international filing date but later than the priority date claimed  "A" later document published after the international filing date or priority date and not in conflict with the application but cited to understand the principle of theory underlying the invention cannot be considered novel or cannot be considered to involve an invention cannot be considered novel or cannot be document is taken alone document of particular relevance; the claimed invention cannot be considered to involve an invention cannot be considered to invention cannot be considered to involve an invention c				
Date of the actual completion of the international search  22 June 1994 (22.06.94)  Date of mailing of the international search report  [Ob. 07.94]				
Name and mailing address of the ISA/AU  Authorized officer				
AUSTRALIA PO BOX 200 WODEN AC	AUSTRALIAN INDUSTRIAL PROPERTY ORGANISATION PO BOX 2000 WODEN ACT 2606 AUSTRALIA			
Facsimile No	Facsimile No. 06 2853929 Telephone No. (06) 2832167			

International application No. PCT/AU 94/00124

ategory*	Citation of document, with indication, where appropriate of the relevant passages	Relevant to Claim No.	
A	US,A, 4306486 (OSWELL) 22 December 1981 (22.12.81) see abstract	1-20	
A	US,A, 4155285 (STOBBE et al) 22 May 1979 (22.05.79) see abstract	1-20	
A	GB, A, 2086549 (DYNAMIT NOBEL A.G.) 12 May 1982 (12.05.82) see abstract	1-20	
•			





International application No. PCT/AU 94/00124

This Annex lists the known "A" publication level patent family members relating to the patent documents cited in the above-mentioned international search report. The Australian Patent Office is in no way liable for these particulars which are merely given for the purpose of information.

	Patent Document Cited in Search Report	Patent Family Member					
us	4155285	BR	7502570	DE	2420862	FR	2269701
GB	2086549	DE	3041149	FR	2493507	US	4438675
					<del></del>	•	
							END OF ANNEX

# This Page is Inserted by IFW Indexing and Scanning Operations and is not part of the Official Record

### **BEST AVAILABLE IMAGES**

Defective images within this document are accurate representations of the original documents submitted by the applicant.

Defects in the images include but are not limited to the items checked:

□ BLACK BORDERS
□ IMAGE CUT OFF AT TOP, BOTTOM OR SIDES
□ FADED TEXT OR DRAWING
□ BLURRED OR ILLEGIBLE TEXT OR DRAWING
□ SKEWED/SLANTED IMAGES
□ COLOR OR BLACK AND WHITE PHOTOGRAPHS
□ GRAY SCALE DOCUMENTS
□ LINES OR MARKS ON ORIGINAL DOCUMENT
□ REFERENCE(S) OR EXHIBIT(S) SUBMITTED ARE POOR QUALITY

## IMAGES ARE BEST AVAILABLE COPY.

OTHER:

As rescanning these documents will not correct the image problems checked, please do not report these problems to the IFW Image Problem Mailbox.

## THIS PAGE BLANK (USPTO)